

FIG. 1

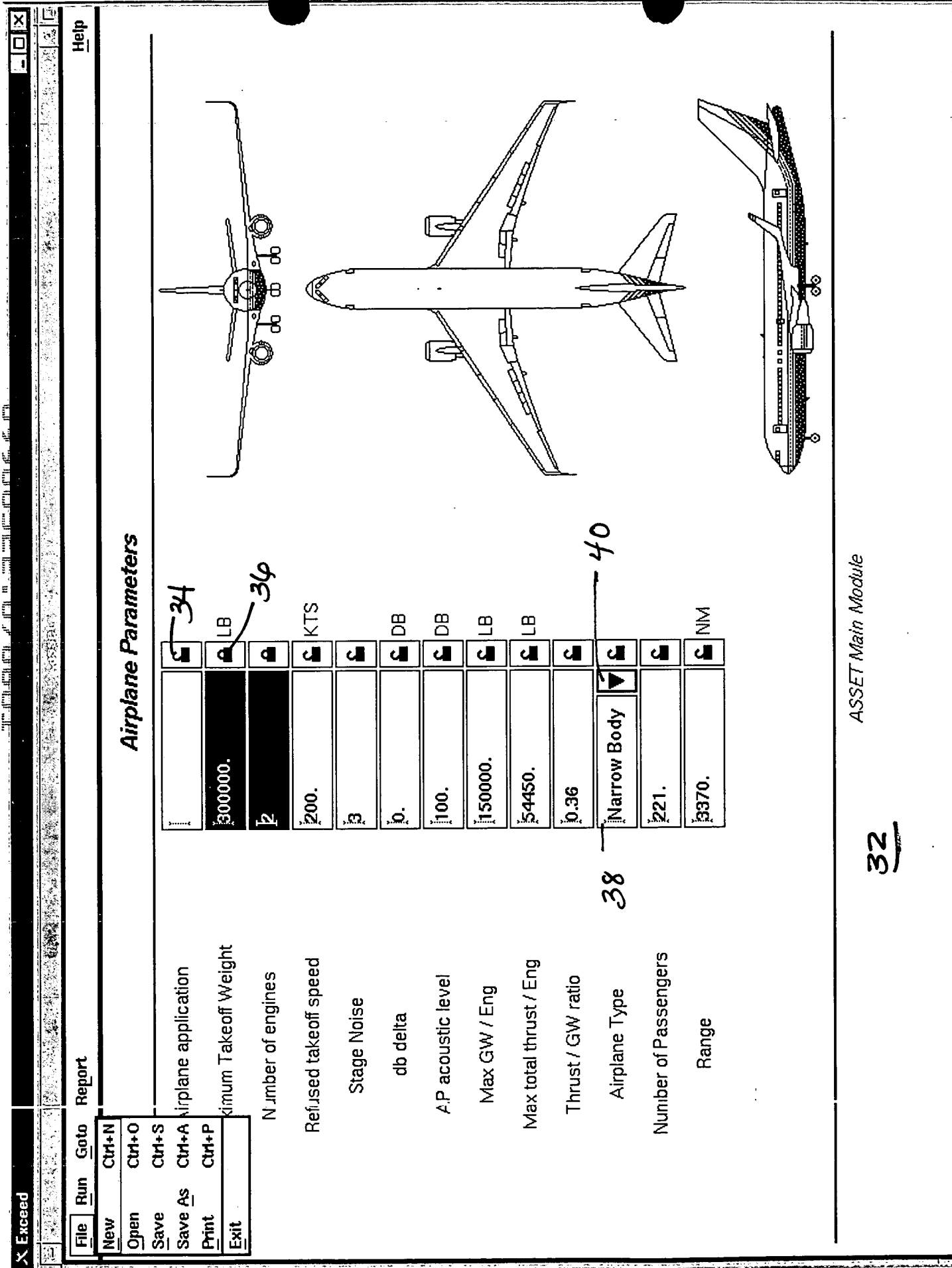
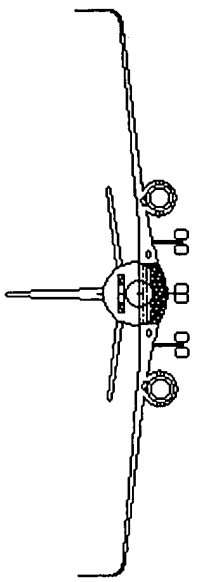
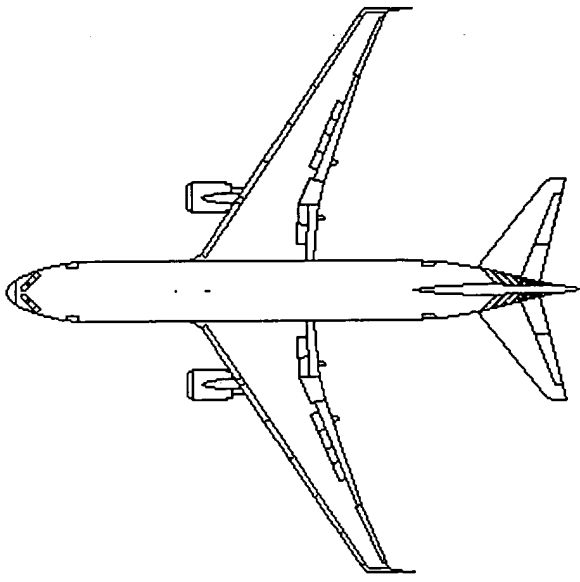
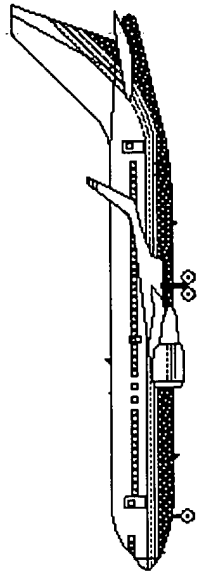


FIGURE 2

Airplane Parameters

Airplane application		
Maximum Takeoff Weight	300000.	LB
Number of engines	2	
Refused takeoff speed	200.	KTS
Stage Noise	3	
db delta	0.	DB
AP acoustic level	100.	DB
Max GW / Eng	150000.	LB
Max total thrust / Eng	54450.	LB
Thrust / GW ratio	0.36	
Airplane Type	Narrow Body	
Number of Passengers	221.	
Range	3370.	NM

ASSET Main Module

FIGURE 3

General:

Fly-by-Wire

☒ 48a TRUE

Frequency Type

Constant

Dual EE Bay

FALSE

Double Voltage

FALSE

RAT Generator?

TRUE

Technology Era

Current

Fuselage Length

154.17

三

Fuselage Diameter

17.27

TF

Number of Passenger Entry/Exit Doors

१

Number of External Power Panels

2.

Fan Diameter

82.96

 \mathbb{Z}

Sweep Angle

35.00

DEG

Wing Span

1525.58

 \mathbb{N}

Horizontal Tail Span

538.95

 \mathbb{Z}









ASSET EPGDS Method

FIGURE 5A

X Exceed		File Run Goto Report		Help	
Configuration					
Body CL to O/B Engine CL	0.00			IN	
Side-of-Body to I/B Engine CL	122.43			IN	
Side-of-Body to O/B Engine CL	0.00			IN	
Dist. along LE I/B Eng. to Side-of-Body	263.32			IN	
Dist along LE O/B Eng. to Side-of-Body	0.00			IN	
Dist. from Fwd. E/E Bay to Front Spar BS	0.00			IN	
Dist. from I/B Eng. to EE Bay	647.27			IN	
Dist. from O/B Eng. to EE Bay	0.00			IN	
Length of Main EE Bay	51.72			IN	
H - Lower Lobe Height	56.02			IN	
W1 Cabin Width	198.98			IN	
W2 Cargo Floor Width	107.78			IN	
Main E/E Bay Volume	257.2			FT^3	
Strut location	Fan				
Accessory location	Core				
ASSET EPGDS Method					

Figure 5B

AC Electrical Load Characterization

Number of Fans	6.0	
Recirculation Fans	2.0	
Number of E/E Cooling Vent Fans	2.0	
Number of E/E Cooling Supply Fans	2.0	
Number of TRUs	3.0	
Number of ACMPs	2.0	
Number of Window/Windshield Heaters	6.0	
Number of Lavatories	3.0	

Number of Wide Body Pumps	Number of Narrow Body Pumps
0.0	6.0
0.0	6.0
0.0	0.0
0.0	0.0

AC Load Summary by Flight Phase

ATA Subsystems	--- Passenger Loading ---				--- Engine Start ---				--- Taxi Out ---			
	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)
21 Air Conditioning	13.72	0.82	13.72	0.82	11.32	0.82	11.32	0.82	11.32	0.82	11.32	0.82
22 Auto Flight	0.68	0.90	0.68	0.90	0.68	0.90	0.68	0.90	0.68	0.90	0.68	0.90
23 Communications	0.64	1.00	0.95	1.00	0.95	1.00	0.95	1.00	2.42	1.00	2.42	1.00
24 Electrical Power	3.38	0.95	3.39	0.95	3.39	0.95	3.39	0.95	3.57	0.95	3.57	0.95
25 Equipment/Furnishings	25.10	1.00	23.24	1.00	23.24	1.00	23.24	1.00	38.93	1.00	38.93	1.00
26 Fire Protection	0.20	1.00	0.20	1.00	0.20	1.00	0.20	1.00	0.20	1.00	0.20	1.00
27 Flight Control	0.07	1.00	0.07	1.00	0.07	1.00	0.07	1.00	0.07	1.00	0.07	1.00
28 Fuel	0.00	1.00	6.08	0.83	6.08	0.83	6.08	0.83	6.08	0.83	6.08	0.83
29 Hydraulic Power System	19.28	0.75	19.28	0.75	19.28	0.75	19.28	0.75	9.28	0.75	9.28	0.75
30 Ice/Rain Protection	4.43	1.00	5.29	1.00	5.29	1.00	5.29	1.00	5.29	1.00	5.29	1.00
31 Instruments	0.30	1.00	0.30	1.00	0.30	1.00	0.30	1.00	0.30	1.00	0.30	1.00
32 Landing Gear	0.12	1.00	0.12	1.00	0.12	1.00	0.12	1.00	0.12	1.00	0.12	1.00
Maximum Flight 1 st Phase Load	116.88	KVA ↔	0.96	PF								

ASSET EPGDS Method

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Figure 7A

FO9040" 22500660

X Exceed

File Run Goto Report

Help

AC Load Summary by Flight Phase

ATA Subsystems	---- Passenger Loading ----				---- Engine Start ----				---- Taxi Out ----			
	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)
32 Landing Gear	<>	<>	<>	<>	<>	<>	<>	<>	<>	<>	<>	<>
33 Lights	10.68	1.00	9.71	1.00	9.32	1.00	9.32	1.00	9.32	1.00	9.32	1.00
34 Navigation	0.89	0.85	0.89	0.85	0.94	0.85	0.94	0.85	0.94	0.85	0.94	0.85
35 Oxygen	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
36 Pneumatics	0.00	1.00	0.23	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
38 Water/Waste	6.36	0.77	1.40	0.83	1.40	0.83	1.40	0.83	1.40	0.83	1.40	0.83
46 Electronic Library	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
49 Airplane Auxiliary Power	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
52 Doors	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
57 Folding Wing	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
73 Engine Fuel Control	0.00	1.00	0.70	0.74	0.70	0.74	0.70	0.74	0.70	0.74	0.70	0.74
74 Ignition	0.00	1.00	0.30	0.33	0.30	0.33	0.30	0.33	0.30	0.33	0.30	0.33

Maximum Flight Phase Load <> 116.88 KVA <> 0.96 PF

ASSET EPGDS Method

FIGURE 7B

AC Load Summary by Flight Phase

ATA Subsystems	--- Take-off & Climb ---			--- Cruise ---			--- Descent & Land ---		
	(kVA)	(PF)	(kVA)	(kVA)	(PF)	(PF)	(kVA)	(PF)	
32 Landing Gear	0.12	1.00	0.12	0.12	1.00	1.00	0.23	1.00	
33 Lights	10.97	1.00	7.73	7.73	1.00	1.00	11.51	1.00	
34 Navigation	1.17	0.88	1.17	1.17	0.88	0.88	1.17	0.88	
35 Oxygen	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	
36 Pneumatics	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	
38 Water/Waste	0.94	0.98	1.14	1.14	0.89	0.94	1.12	0.94	
46 Electronic Library	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
49 Airplane Auxiliary Power	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	
52 Doors	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	
57 Folding Wing	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	
73 Engine Fuel Control	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	
74 Landing Gear	0.00	1.00	0.00	0.00	1.00	1.00	0.00	1.00	

Maximum Flight Phase Load <> 116.88 KVA <> 0.96 PF

ASSET EPGDS Method

Figure 7c

AC Load Summary by Flight Phase

ATA Subsystems	--- Take-off & Climb ---						--- Cruise ---						--- Descent & Land ---					
	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)		
57 Forward wing	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
73 Engine Fuel Control	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
74 Ignition	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
75 Air	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
76 Engine Controls	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
77 Engine Indicating	<>	0.02	<>	1.00	<>	0.02	<>	1.00	<>	0.02	<>	1.00	<>	0.02	<>	1.00		
78 Exhaust	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
79 Oil	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
80 Starting	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00	<>	0.00	<>	1.00		
Flight Phase Subtotals	<>	101.63	<>	0.96	<>	100.16	<>	0.96	<>	65.71	<>	0.96	<>	65.71	<>	0.93		
Error/Growth Factor (15%)	<>	15.24	<>	0.96	<>	15.02	<>	0.96	<>	9.86	<>	0.96	<>	9.86	<>	0.93		
Flight Phase Totals	<>	116.88	<>	0.96	<>	115.19	<>	0.96	<>	75.57	<>	0.96	<>	75.57	<>	0.93		

Maximum Flight Phase Load <>	116.88	KVA <>	0.96	PF
------------------------------	--------	--------	------	----

ASSET EPGDS Method

Essential AC Loads

	Quantity	Load per Unit	Totals
Number of Upper Recirculating Fans	2.0	1.28 KVA	15.38 KVA
Number of Lower Recirculating Fans	0.0	1.98 KVA	
Number of E/E Cooling Supply Fans	2.0	3.20 KVA	
Number of E/E Cooling Vent Fans	2.0	3.20 KVA	
Total Fan Load			
Number of Hydraulic ACMP Pumps	2.0	6.41 KVA	23.30 KVA
Number of Fuel Boost Pumps	6.0	1.75 KVA	
Number of Fuel Override Pumps	0.0	4.66 KVA	
Total Pump Load			
Passenger Load			7.08 KVA
Baseline Flight & Electronics Total Load			13.10 KVA
Baseline Flight & Electronics, Ice & Rain			
Baseline Flight & Electronics, Electronics			
Subtotal of Essential Loads			58.86 KVA
General Feeder Loss			4.12 KVA
Total of Essential Loads			62.98 KVA

ASSET EPGDS Method

File		Run		Goto		Report		Help	
				Next					
				Previous					
				Back					
				Airplane Parameters					
				NACELLE					
				EPGDS					
				Leading Edge					
				Floor Beam					
				Refueled takeoff speed					
				Stage Noise					
				db delta					
				AP acoustic level					
				Max GW / Eng					
				Max total thrust / Eng					
				Thrust / GW ratio					
				Airplane Type					
				Number of Passengers					
				Range					

Airplane Parameters

Weight	300000.	LB
Configuration	Full AC	
Loads	Essential AC Loads	
Architecture	DC Electrical Load Characterization	KTS
Generation	Standby DC Loads	
Distribution	DC Load Summary by Flight Phase	
System Attributes	IFE	
Weight Summaries	0.	DB
	100.	DB
	150000.	LB
	54450.	LB
	0.36	
	Narrow Body	
	221.	
	3370.	NM

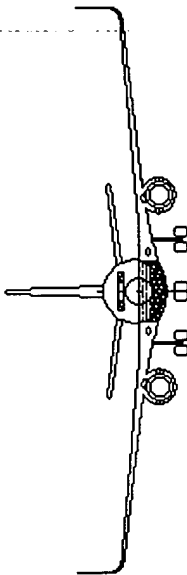
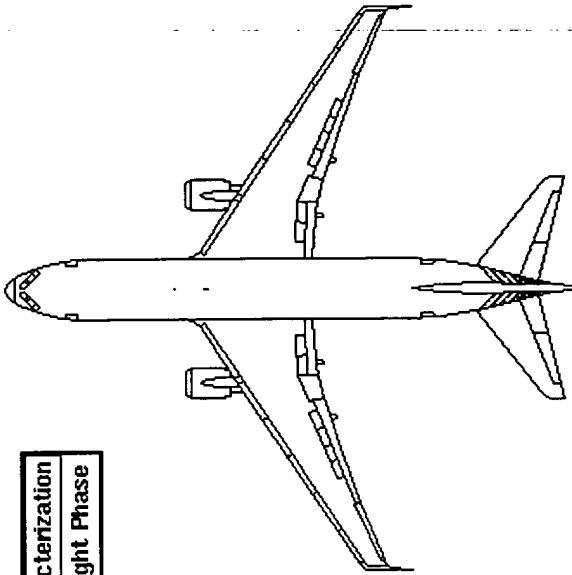
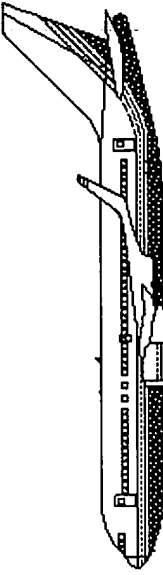








Figure 9

DC Electrical Load Characterization

Number of Main Landing Gear Wheels	4.0	
Number of APU Generators	1.0	
Number of Doors	6.0	
Number of Tanks	3.0	

ASSET EPGDS Method

Figure 10

ATA Subsystems

ATA Subsystems	Pass		Engine		Take-Off		Descent	
	Loading	Start	Taxi-Out	Cruise	& Land			
	(Amps)	(Amps)	(Amps)	(Amps)	(Amps)			
21 Air Conditioning	13.40	14.60	15.01	15.14	15.14			
22 Auto Flight	9.83	9.83	9.83	10.42	10.75			
23 Communications (IFE, AVOD)	6.74	6.56	7.69	6.90	7.13			
24 Electrical Power	3.36	2.27	2.27	2.27	2.27			
25 Equipment/Furnishings	18.22	18.04	18.04	18.04	18.04			
26 Fire Protection	0.54	0.54	0.54	0.54	0.54			
27 Flight Control	0.07	0.07	0.07	0.07	0.07			
28 Fuel	6.51	1.21	1.21	1.21	1.21			
29 Hydraulic Power System	0.80	0.80	0.80	0.80	0.80			
30 Ice/Rain Protection	3.20	5.30	12.42	12.42	5.40			
31 Instruments	36.81	36.70	36.70	36.70	36.70			
Maximum Flight Phase Direct Current Load					139.90			
					AMPS			

ASSET EPGDS Method

DC Load Summary by Flight Phase

ATA Subsystems	Pass		Engine		Take-Off		Cruise		Descent	
	Loading	(Amps)	Start	(Amps)	Taxi-Out	(Amps)	& Climb	(Amps)	& Land	(Amps)
31 Instruments	<>	36.81	<>	36.70	<>	36.70	<>	36.70	<>	36.70
32 Landing Gear	<>	3.69	<>	3.69	<>	3.67	<>	3.59	<>	4.07
33 Lights	<>	15.84	<>	15.77	<>	16.38	<>	19.47	<>	16.05
34 Navigation	<>	1.99	<>	1.95	<>	2.45	<>	2.45	<>	2.45
35 Oxygen	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00
36 Pneumatics	<>	4.07	<>	4.07	<>	4.07	<>	4.07	<>	4.07
38 Water/Waste	<>	2.07	<>	1.53	<>	1.53	<>	2.07	<>	2.07
46 Electronic Library	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00
49 Airplane Auxiliary Power	<>	1.20	<>	1.20	<>	1.20	<>	1.20	<>	0.00
52 Doors	<>	1.00	<>	1.50	<>	1.50	<>	1.50	<>	1.50
57 Folding Wing	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00
Maximum Flight Phase Direct Current Load		39.90								

AMPS

AMPS

ASSET EPGDS Method

Figure 11B

ATA Subsystems	Pass		Engine		Taxi-Out		Take-Off & Climb		Cruise		Descent & Land		
	Loading	(Amps)	Start	(Amps)	(Amps)	(Amps)	(Amps)	(Amps)	(Amps)	(Amps)	(Amps)		
52 Doors	<>	1.00	<>	1.50	<>	1.50	<>	1.50	<>	1.50	<>	1.50	
57 Folding Wing	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	
73 Engine Fuel Control	<>	0.00	<>	0.07	<>	0.37	<>	0.37	<>	0.37	<>	0.37	
74 Ignition	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	
75 Air	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	
76 Engine Controls	<>	1.12	<>	1.12	<>	0.65	<>	0.65	<>	0.65	<>	0.65	
77 Engine Indicating	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	
78 Exhaust	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.60	
79 Oil	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	
80 Starting	<>	0.00	<>	3.20	<>	0.00	<>	0.00	<>	0.00	<>	0.00	
Flight Phase Totals	<>	130.46	<>	130.02	<>	136.40	<>	139.90	<>	129.41	<>	139.53	
Maximum Flight Phase Direct Current Load												139.90	AMPS

ASSET EPGDS Method

FIGURE 11C

IFE

Technology Era Constant	1.000	
System Factor	0.700	
Airflow Constant	156.0	CFM/KVA
Fan performance coefficient	0.00196522	KVA/CFM
IFE Power Factor	0.98	
IFE Utilization Factor	100.0	
IFE Load	0.0	KVA

ASSET EPGDS Method

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FIGURE 13

AP acoustic level

Max GW / Eng

Max total thrust / Eng

Thrust / GW ratio

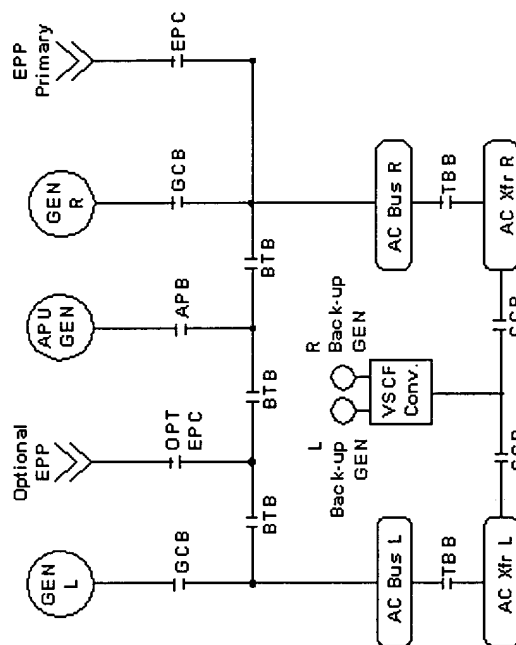
Airplane Type

Number of Passengers

Range

Main/Backup AC System

Twin, Fly-by-Wire, Isolated Architecture



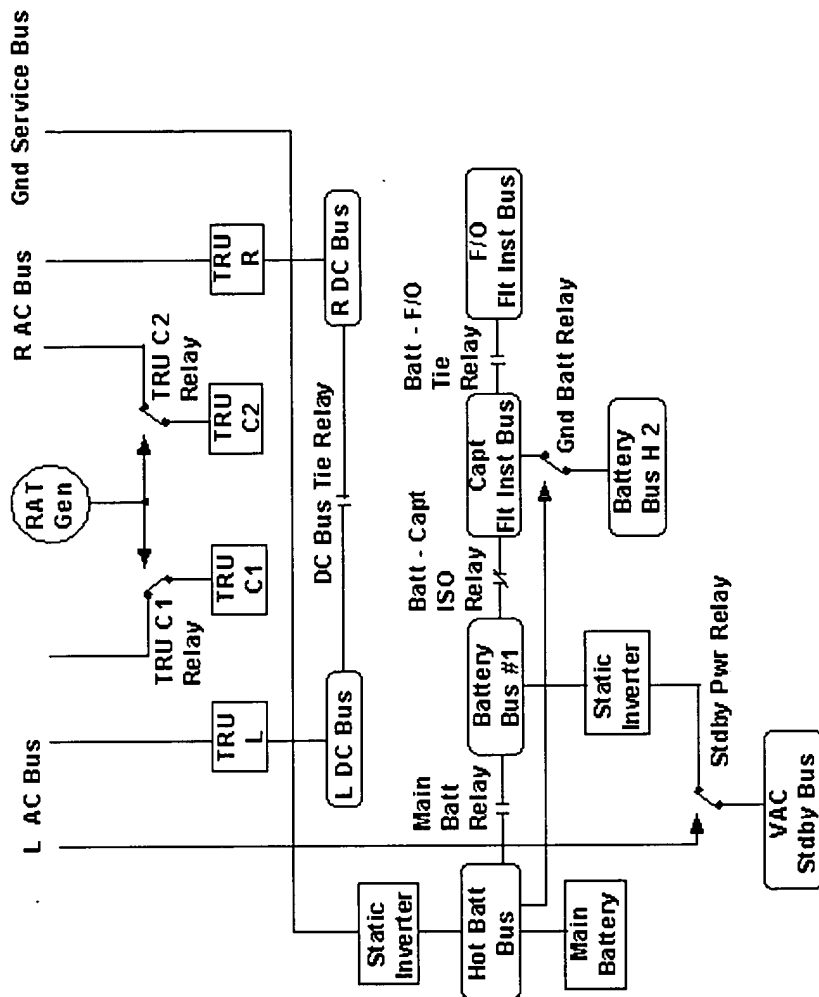
If selected, HMG Backup System not reflected in graphics.

Main Generator Type	IDG		
Optional External Power Panel	<input checked="" type="checkbox"/> TRUE		
Number of Generators per Engine	1		
In-Flight APU Generator	<input checked="" type="checkbox"/> In-flight operable		
Backup Generator	<input checked="" type="checkbox"/> VSCF: Stand-Alone Converter		
HMG Option?	<input type="checkbox"/> FALSE		

ASSET EPGDS Method

Figure 15

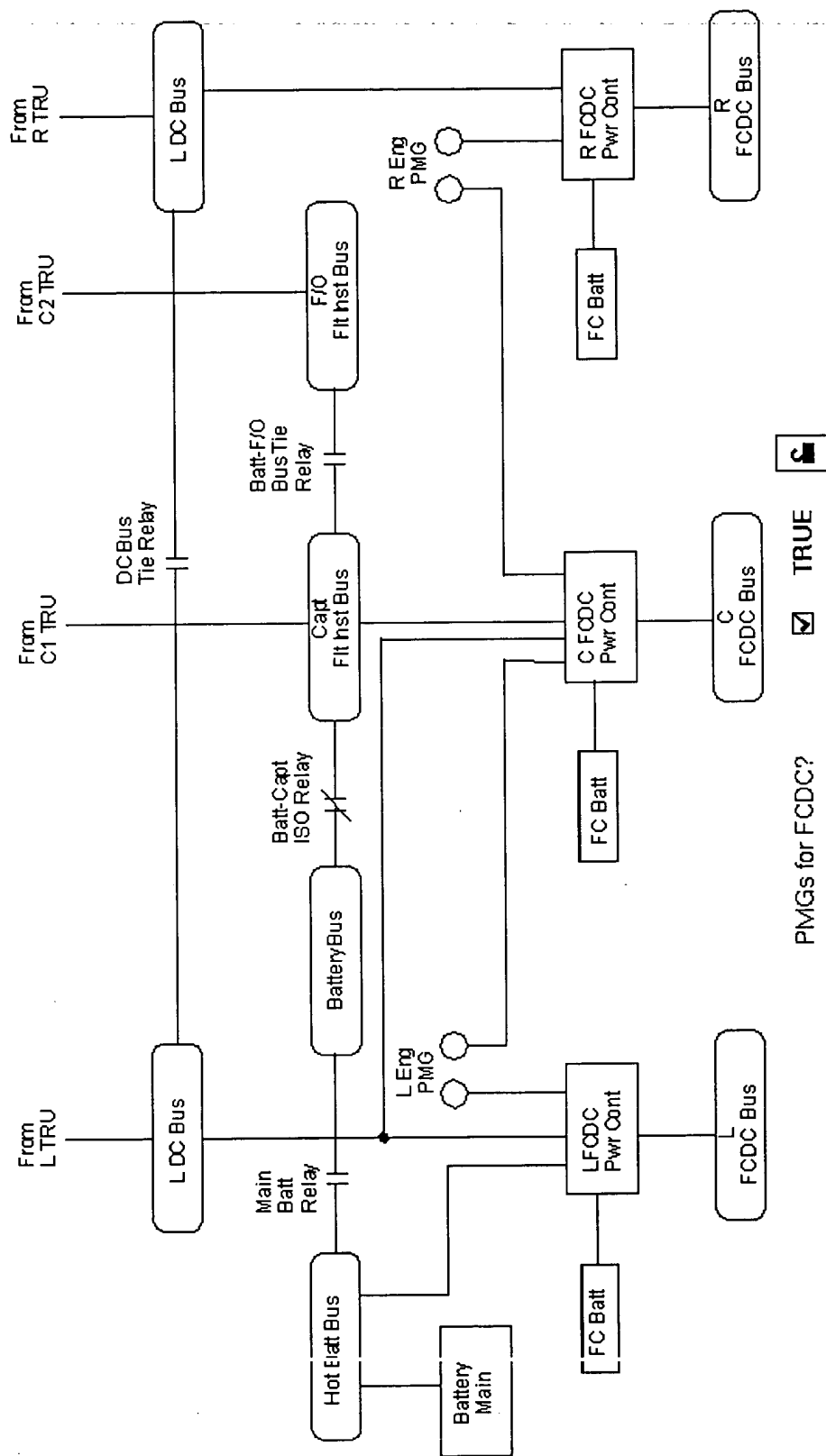
Twin, Fly-by-Wire



ASSET EPGDS Method

Figure 16

Twin, Fly-by-Wire



ASSET EPGDS Method

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Figure 17

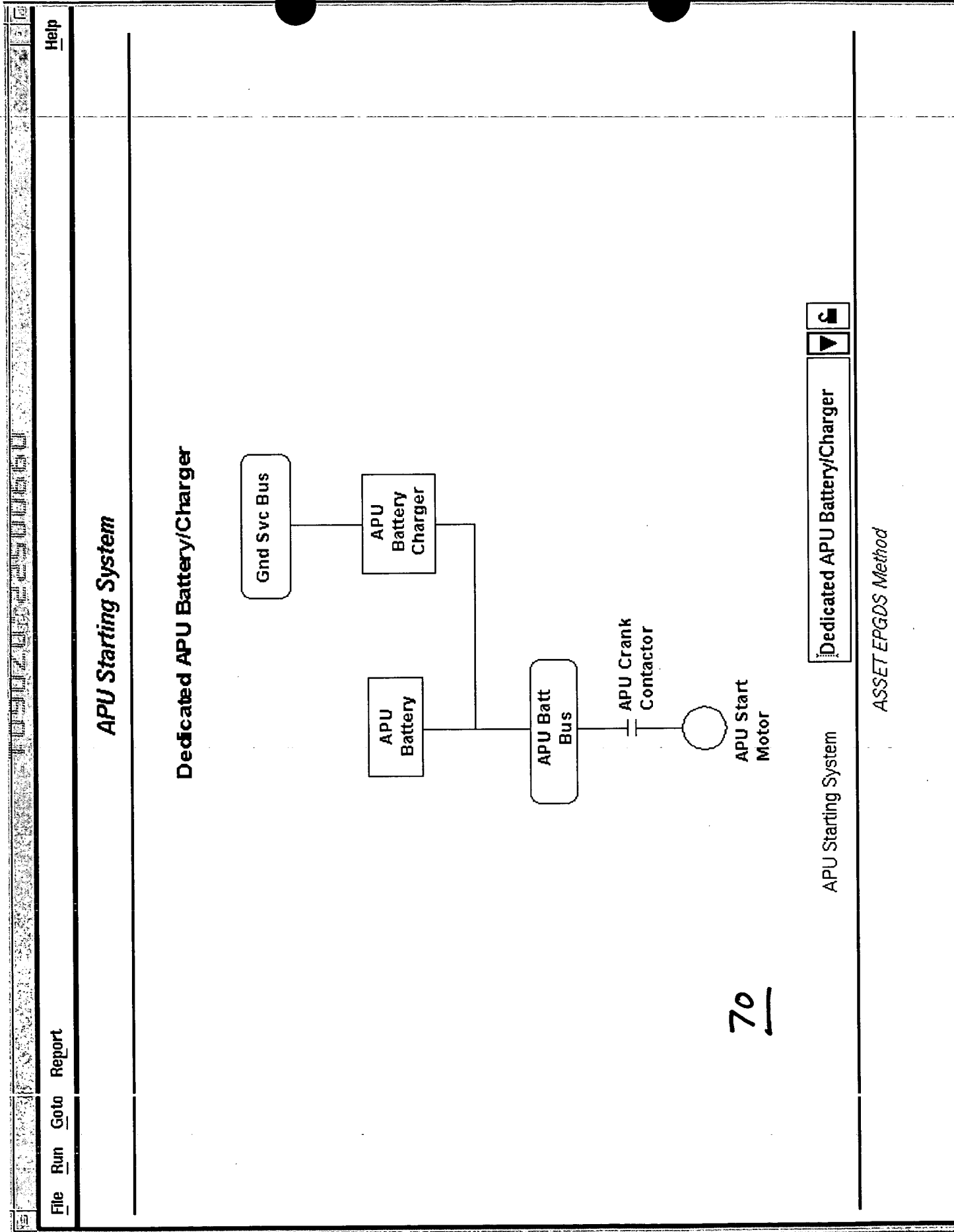
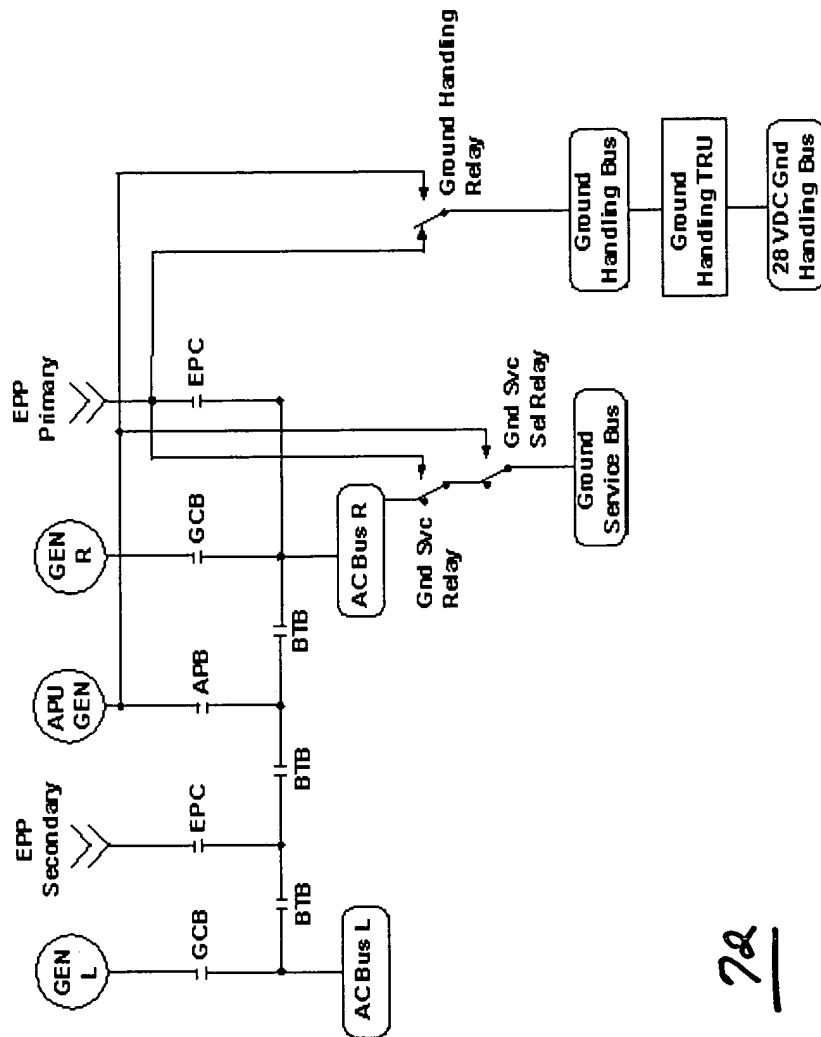


Figure 18

Twin, Fly-by-Wire and Non-Fly-by-Wire



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ASSET EPGDS Method

FIGURE 19

File		Run		Goto		Report		Next		Previous:		Back		Help	
Airplane Parameters															
Airplane Parameters		NACELLE		EPGDS		Leading Edge		Floor Beam		Refused takeoff speed		Stage Noise		db delta	
on		Weight		Configuration		Loads		Architecture		Generation		Distribution		System Attributes	
300000.		2		2		2		2		2		2		2	
LB		LB		LB		LB		LB		LB		LB		LB	
AC Power Generation		APU Generator		Emergency Power Generation		Generator Control Units		Back Up AC Power		Transformer Rectifier Unit (TRU)		Batteries and Battery Chargers		Flight Control DC Power	
Transformers		0.36		Narrow Body		221.		3370.		NM					
Max total thrust / Eng		Thrust / GW ratio		Airplane Type		Number of Passengers		Range							

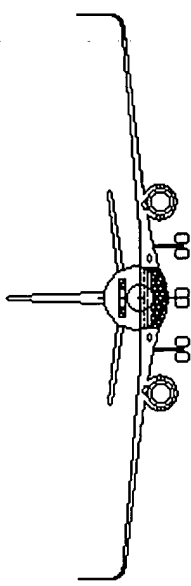
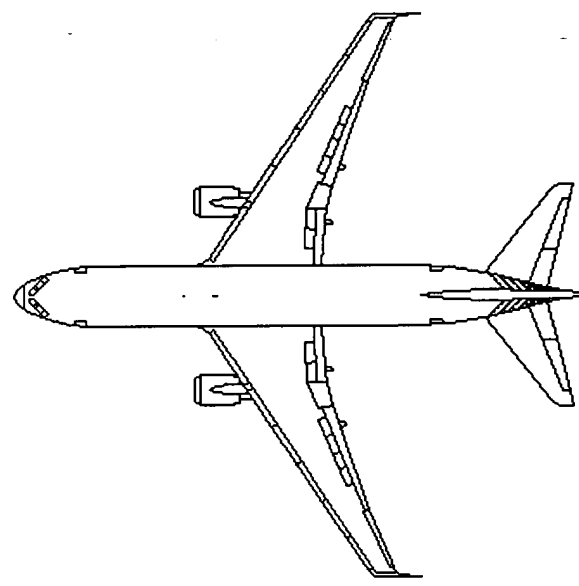

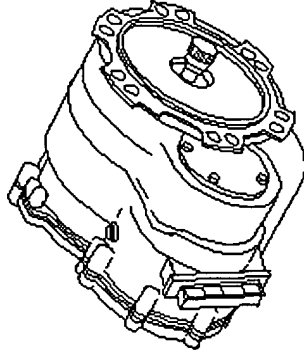




FIGURE 20

Main Converter Unit Weight

60



Motor Controller Weight LB

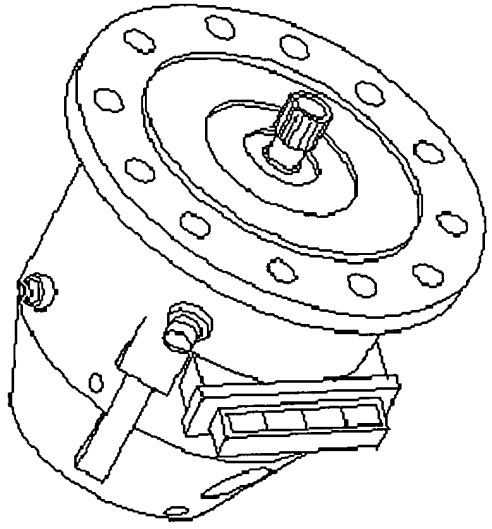
[illegible]

ASSET EPGDS Method

 力 L

FIGURE 21

APU Generator



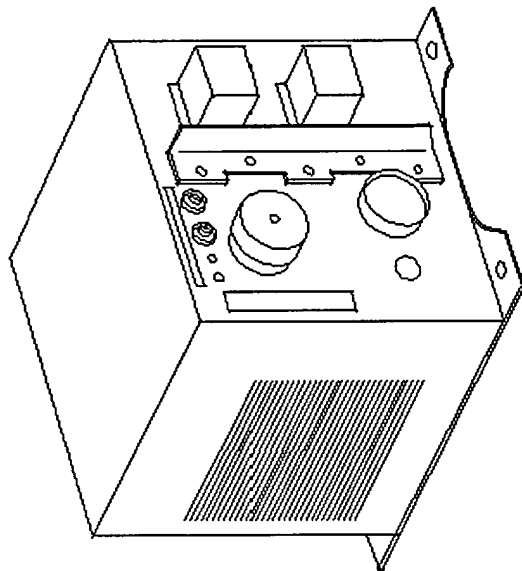
In-Flight Operable APU	<input checked="" type="checkbox"/>	TRUE	<input type="checkbox"/>
APU Generator Capacity	50.0		KVA
APU Generator Weight	64.2		LB
Number of APU Generators	1.0		

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ASSET EPGDS Method

Figure 22

Generator Control Units



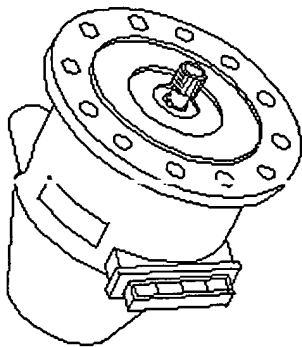
78

Unit Size	Main AC		<>	3.0	APU		3.0	<>	3.0	RAT		3.0	MCU
Unit Weight		5.0	<>	5.0		5.0	<>	5.0			5.0	5.0	LB

ASSET EPGDS Method

Back Up AC Power

VSCF



Generator Type

Capacity

Cooling Method

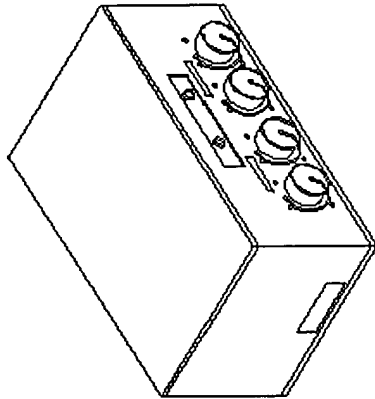
Input speed

Generator Weight

Converter Configuration

Converter Weight

PMGS



VSCF System

18.8

Air

12000.0

38.1

KVA

RPM

87

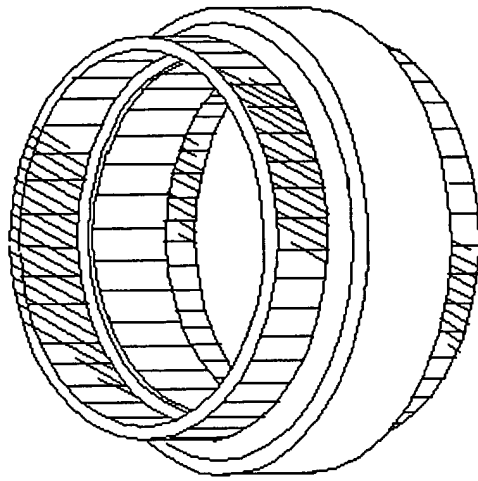
Number/Engine

PMG Configuration

PMG Unit Weight

Stand Alone Converter

43.1 LB



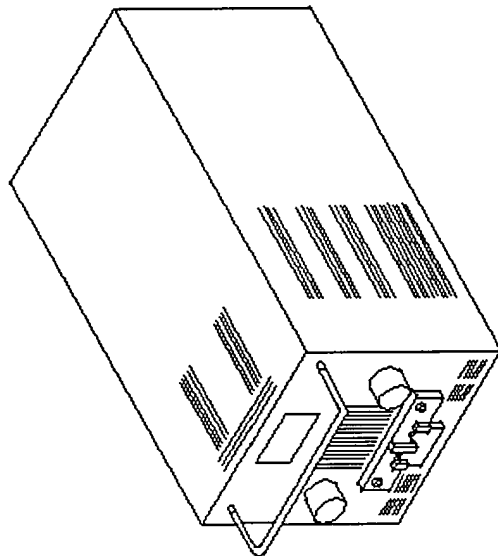
5





Integrated with Back Up Generator

2.5

ASSET EPGDS Method

Transformer Rectifier Unit (TRU)

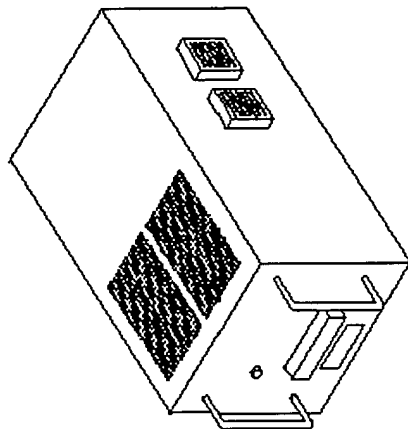
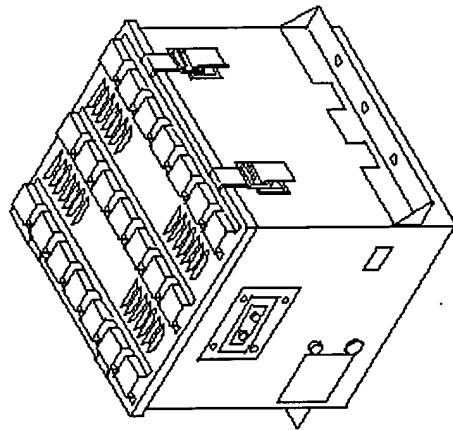


Number of TRUs	3.0	
DC Output	150.0	
Efficiency	92.0	
TRU Weight	10.7	
		AMPS
		%
		LB



ASSET EPGDS Method

FIGURE 25

Batteries and Battery Chargers





Batteries

Nominal Capacity	47.0	
Battery Weight	106.0	

MAIN Battery

Output Capacity	60.0	AMPS
Battery Charger Weight	13.0	LB

Battery Chargers

Nominal Capacity	47.0	
Battery Weight	106.0	

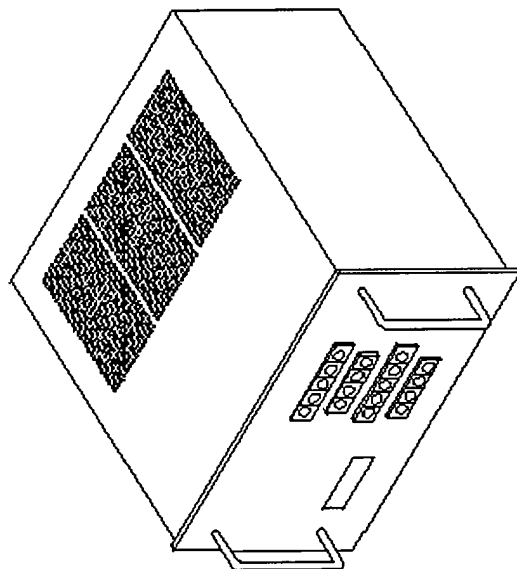
Output Capacity	60.0	AMPS
Battery Charger Weight	13.0	LB

ASSET EPGDS Method

FIGURE 26

Flight Control DC Power

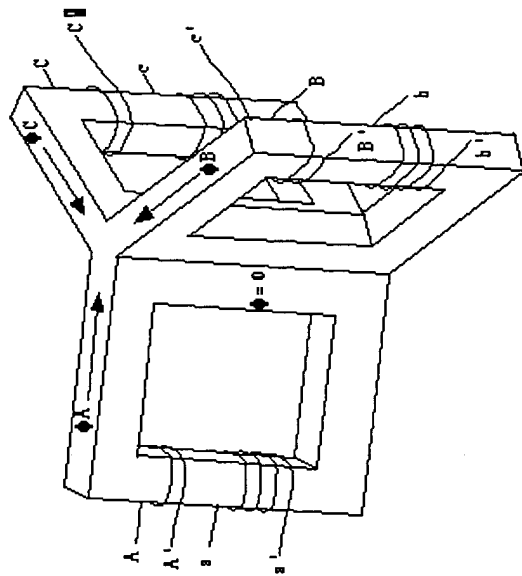
Power Supply Assemblies (PSAs)



Output Power	900.0	WATTS	Number of Dedicated Batteries	3	LB
Converter Architecture	Dual Converter		PSA Battery Unit Weight	4.3	LB
PSA Cabinet Weight	35.0	LB			

ASSET EPGDS Method

FIGURE 27







Step-Up Transformer Capacity

Step-Up Transformer

Step-Down Transformer Capacity

Step Down Transformer Weight

0.0		KVA
0.0		LB
0.0		KVA
0.0		LB






ASSET EPGDS Method

FIGURE 28


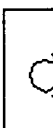
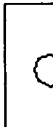

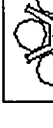



Feeder Configuration

Show Data for:

MAIN

<i>Feeder 1:</i>	<>	3-Wire w/Ntrl	▼	
<i>Feeder 2:</i>	<>	2 3-Wire w/Ntrl	▼	
<i>Feeder 3:</i>	<>	2 3-Wire w/Ntrl	▼	
<i>Feeder 4:</i>	<>	2 3-Wire w/Ntrl	▼	
<i>Feeder 5:</i>	<>	None	▼	

Bundle Cross-Sections

	3-Wire		3-Wire w/Spkr		2 3-Wire		2 3-Wire w/Ntrl		2 3-Wire w/Ntrl w/Spkr		6-Wire w/Ntrl		6-Wire w/Spkr		Blank
-------------------------------------------------------------------------------------	---------------	-------------------------------------------------------------------------------------	----------------------	-------------------------------------------------------------------------------------	-----------------	-----------------------------------------------------------------------------------	------------------------	-----------------------------------------------------------------------------------	-------------------------------	---------------------------------------------------------------------------------------	----------------------	---------------------------------------------------------------------------------------	----------------------	---------------------------------------------------------------------------------------	--------------

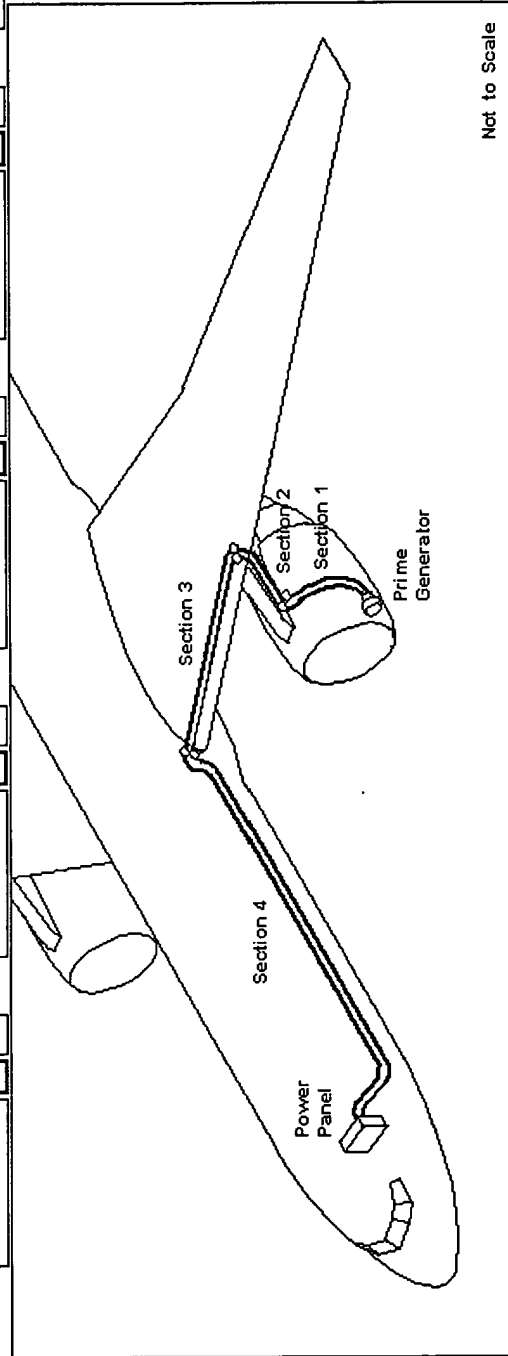
ASSET EPGDS Method

Figure 30

Feeder Diagram

Show Data for: **MAIN**

	Feeder 1		Feeder 2		Feeder 3		Feeder 4		Feeder 5	
Material	CU		CU		AL		AL		None	
Gage	3/0		4		2		4		None	
	Neutral 1		Neutral 2		Neutral 3		Neutral 4		Neutral 5	
Material	CU		CU		AL		AL		None	
Gage	1/0		8		6		8		None	



ASSET EPGDS Method

FIGURE 31

Feeder Analysis

Show Data for: MAIN

	Feeder 1	Feeder 2	Feeder 3	Feeder 4	Feeder 5
Phase Current	260.9	130.4	130.4	130.4	0.0
Feeder Temperature Rise	49.7	59.2	61.6	84.6	0.0
Bundle Derating	1.000	1.000	1.000	1.000	1.000
Sizing Altitude	41000	8000	8000	8000	8000
Altitude Derating	1.757	1.112	1.112	1.112	1.112
Ambient Temperature	149.0	93.0	93.0	57.0	57.0
Feeder Temperature	236.4	158.8	161.5	151.1	57.0
Max Wire Temperature	260.0	260.0	175.0	175.0	260.0
Temperature Margin	23.6	101.2	13.5	23.9	203.0
Feeder Length	6.84	14.38	26.33	38.40	0.00

Maximum Voltage Drop	10.000	VOLTS
Total Voltage Drop	7.757	VOLTS
Voltage Drop Margin	2.243	VOLTS

ASSET EPGDS Method

Wire Type & Weight




Show Data for: MAIN

Wire Type, Feeder 1:	<>	BMS 13-60 Type 7	▼	11.2	LB	Feeder 1:	<>	11.2	LB
Wire Type, Neutral 1:	<>	BMS 13-60 Type 7	▼	2.4	LB	Neutral 1:	<>	2.4	LB
Wire Type, Feeder 2:	<>	BMS 13-60 Type 7	▼	12.5	LB	Feeder 2:	<>	12.5	LB
Wire Type, Neutral 2:	<>	BMS 13-60 Type 7	▼	1.7	LB	Neutral 2:	<>	1.7	LB
Wire Type, Feeder 3:	<>	BMS 13-35 Type 1	▼	15.2	LB	Feeder 3:	<>	15.2	LB
Wire Type, Neutral 3:	<>	BMS 13-35 Type 1	▼	2.3	LB	Neutral 3:	<>	2.3	LB
Wire Type, Feeder 4:	<>	BMS 13-35 Type 1	▼	14.7	LB	Feeder 4:	<>	14.7	LB
Wire Type, Neutral 4:	<>	BMS 13-35 Type 1	▼	2.3	LB	Neutral 4:	<>	2.3	LB
Wire Type, Feeder 5:	<>	BMS 13-60 Type 22	▼	0.0	LB	Feeder 5:	<>	0.0	LB
Wire Type, Neutral 5:	<>	BMS 13-60 Type 22	▼	0.0	LB	Neutral 5:	<>	0.0	LB
TRU Feeder Weight						6.4	LB		
Total Wire Weight						68.8	LB		

ASSET EPGDS Method

Panel Technology Selection

Technology Factors:

Backplane	1.00	
ELMS	1.00	
Other	1.00	

ASSET EPGDS Method

System Acquisition Costs

System Acquisition Cost, Base Year (per fleet)	0.	DOLLARS
System Support Equipment Cost, Base Year (per fleet)	0.	DOLLARS
System Initial Training Cost, Base Year (per fleet)	0.	DOLLARS
System Acquisition Cost per Airplane per Year	4078.	DOLLARS

ASSET EPGDS Method

FIGURE 37

Fuel Costs

Fuel Cost per Gallon, Base Year	\$0.49	\$	DOLLARS
Lbs Fuel Burned / Flight Hour / Lb Additional Weight	\$0.0310	\$	HRS^-1
System Weight (per airplane)	\$221.0	\$	LB
System Direct Horsepower Requirement (per airplane)	\$0.	\$	HP
System Drag Horsepower Requirement (per airplane)	\$0.	\$	HP
System Cooling Horsepower Requirement	\$0.	\$	HP
System Pound of Fuel per Block Trip (per airplane)	\$0.	\$	LB
Average Fuel Inflation Rate Beyond Present Year	\$0.035	\$	%
Fuel Cost (NPV of Life Cycle Cost)	\$15800.	\$	DOLLARS
Fuel Cost per Airplane per Year	\$1860.	\$	DOLLARS

ASSET EPGDS Method

FIGURE 38

Spares Costs

Cost / Spare Unit, Base Year	270000.	DOLLARS
Spares Holding Factor	0.12	%
Shop Turnaround Time in Days	34.0	DAYS
Main Base Fill Rate (must be less than 1)	0.95	
Mean Time Between Unscheduled Removals	12000.	HRS
Mean Time Between Overhauls	0.	HRS
Number of Spares Required	5.	
Initial Spares Cost	1350000.	DOLLARS
Spares Holding Cost (NPV of Life Cycle Cost)	1787786.	DOLLARS
Spares Cost (NPV of Life Cycle Cost)	3137786.	DOLLARS
Spares Cost per Airplane per Year	9478.	DOLLARS

ASSET EPGDS Method

FIGURE 39

Line Maintenance Costs

Direct Labor Rate per Hour	21.00	DOLLARS/HOUR
Maintenance Labor Burden Factor	2.4	
Mean Time Between Unscheduled Removals	12000.	HRS
Line Labor Hours Required per Removal	2.0	HRS
Line Labor Hours per Maintenance Action (Non-Removal)	0.5	HRS
Maintenance Actions per 1000 Flight Hours (Non-Removal)	0.50	HRS^-1
Line Maintenance Cost (NPV of Life Cycle Cost)	73673.	DOLLARS
Line Maintenance Cost per Airplane per Year	223.	DOLLARS

ASSET EPGDS Method


Figure 40

Shop Maintenance Costs

Direct Labor Rate per Hour	21.00	DOLLARS/HOUR
Maintenance Labor Burden Factor	2.4	
Mean Time Between Unscheduled Removals	12000.	HRS
Main Generator Mean Time Between Failures	26000.	HRS
Mean Time Between Overhauls	0.	HRS
Shop Labor Man-Hours per Unconfirmed Failure (Test Time)	8.0	HRS
Shop Labor Man-Hours per Failure (Repair and Test)	48.0	HRS
Shop Labor Hours per Overhaul	0.0	HRS
Average Shop Material Cost per Failure, base year	67500.	DOLLARS
Overhaul Materials Cost per Overhaul	0.	DOLLARS
Shop Maintenance Cost (NPV of Life Cycle Cost)	6819057.	DOLLARS
Shop Maintenance Cost per Airplane per Year	20597.	DOLLARS

ASSET EPGDS Method

Schedule Interruption Costs

		DOLLARS/HOUR
Average Delay Cost per Delay Hour	10300.	
Average Cancellation Cost per Cancellation	51000.	
Average Air Turnback Cost per Turnback	36700.	DOLLARS
Average Diversion Cost per Diversion	43000.	DOLLARS

Number of Delays per 100 Departures	0.0030	
Average Delay Time (Hours)	1.70	HRS
Number of Cancellations per 100 Departures	0.0001	
Number of Air Turnbacks per 100 Departures	0.0002	
Number of Diversions per 100 Departures	0.0000	

Schedule Interruptions Cost (NPV of Life Cycle Cost)	\$939999.	\$	DOLLARS
Schedule Interruptions Cost per Airplane per Year	1492.	\$	DOLLARS

ASSET EPGDS Method

FIGURE 43

Dependability Cost Summary

	NPV of Life Cycle Cost	Per Airplane per Year	
Line Maintenance Cost	73673.	223.	DOLLARS
Shop Maintenance Cost	6819057.	20597.	DOLLARS
Scheduled Maintenance Cost	1237712.	3739.	DOLLARS
Schedule Interruptions Cost	493999.	1492.	DOLLARS
Spares Cost	3137786.	9478.	DOLLARS
Fuel Cost	615800.	1860.	DOLLARS
Dependability Cost	12378028.	37388.	DOLLARS

ASSET EPGDS Method

Figure 44

Reliability Inputs

Average Flight Hours per Flight

3.40



LRU MTBF's

Main Generator MTBF

26000.



APU Generator MTBF

20000.



VSCF Backup Generator MTBF

20000.



Generator Control Unit (GCU) MTBF

250000.



Backup Converter MTBF

15000.



Generator Control Breaker (GCB) MTBF

3000000.



IFSD Rates (per 1000 flight hours)

Engine In-flight Shutdowns per 1000 hours

0.010



HRS^-1

APU In-flight Shutdowns per 1000 hours

0.200



HRS^-1

Failure to Start Probabilities

APU No-Start Probability

0.010



Probability of RAT Unavailable when Required

3.2e-03



Other Failure Rates (per flight hour)

Ram Air Turbine MTBF

10000.



RAT Gen. Control Unit MTBF

40000.



Permanent Magnet Generator (PMG) MTBF

420000.



Main and APU Battery MTBF

25000.



Main and APU Battery Charger MTBF

30000.



Rate of Other Channel Faults

1.2e-05



HRS^-1

Main Generator Shaft Shear Rate

3.0e-06



HRS^-1

Backup Generator Shaft Shear Rate

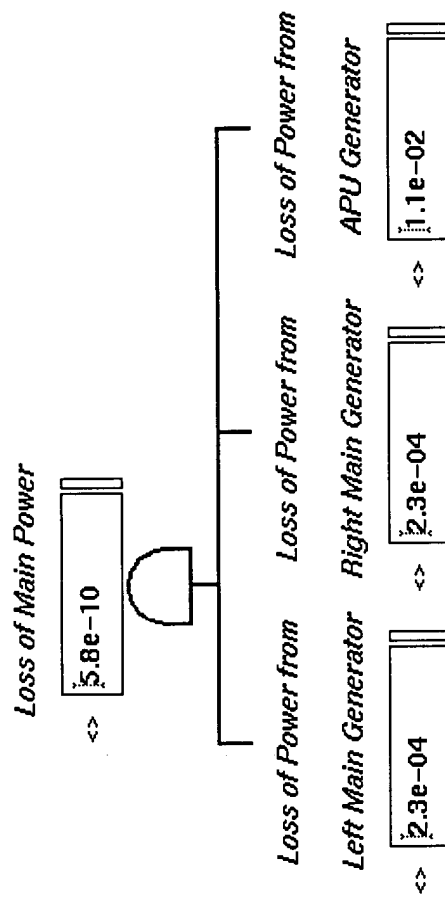
1.2e-05



HRS^-1

ASSET EPGDS Method

Main Power Reliability



Loss of Main Power

5.8e-10

5.8e-10



Loss of Power from

Loss of Power from

Loss of Power from

Left Main Generator

Bight Main Generator

APU Generator

$$2.3e-04$$

2.3e-04

Y

2.3e-04

 Δ

1.1e-02

D

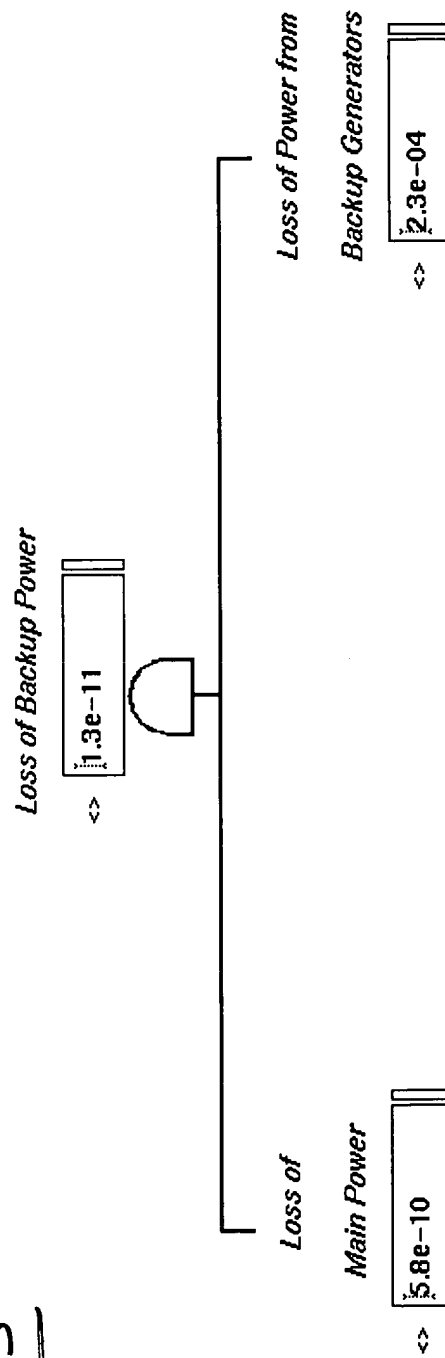
ASSET EPGDS Method

***** NOTE: This is a Read-Only screen *****

FIGURE 47

Backup Power Reliability

58

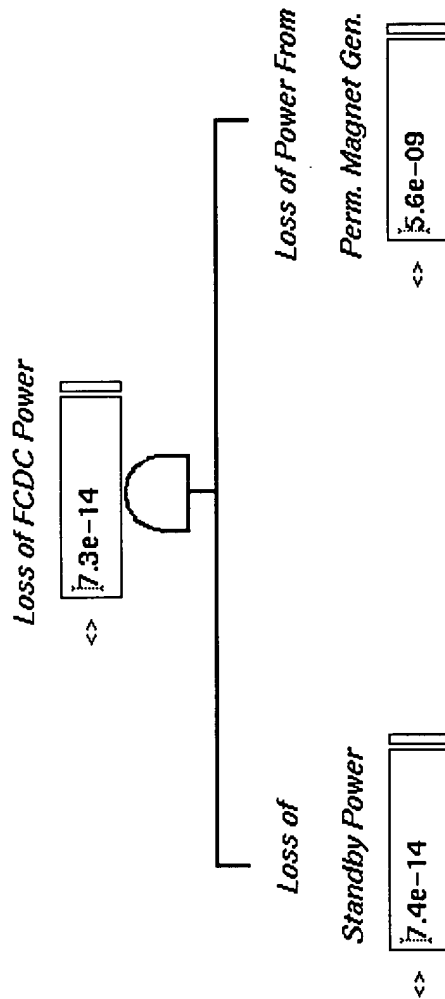


AND Gate OR Gate



***** NOTE: This is a Read-Only screen *****

FIGURE 48

FCDC Power Reliability



8.8

 AND Gate
  OR Gate

***** NOTE: This is a Read-Only screen *****

Figure 50

Maintenance Times

	Unscheduled Removals	Servicing	Alignment & Adjustment
<i>Frequency (Flight Hours)</i>			
Mean Time Between Unscheduled Removals	12000.	750.	4000.
Maintenance Interval			
<i>Maintenance Corrective Times (Flight Hours)</i>			
Main Generator Unscheduled Removal Access Time	0.50	0.50	0.50
Main Generator Unscheduled Removal Fault Isolation Time	0.50		
Repair / Remove & Replace Time	2.00		
Main Generator Unscheduled Removal Servicing Time	1.00	2.00	
Main Generator Unscheduled Removal Alignment & Adjustment Time	0.50		0.60
Main Generator Unscheduled Removal Checkout / Verification Time	1.00		1.00
Main Generator Unscheduled Removal Closing Up Time	0.60	0.60	0.60
Main Generator Unscheduled Removal Mean Corrective Time	6.	3.	3.

ASSET EPGDS Method

90

Preparation Times

	Unscheduled	Servicing	Alignment & Adjustment
<i>Maintenance Preparation Times (Flight Hours)</i>			
Main Generator Unscheduled Removal Maintenance Coordination Time	0.10	0.10	0.10
Main Generator Unscheduled Removal Dispatch Delay Time	0.20		
Main Generator Unscheduled Removal Airplane Ferrying Time	1.00		
Main Generator Unscheduled Removal Supply Delay Time	0.	0.10	
Main Generator Unscheduled Removal Spares & Equipment Issuing Time	0.50		0.50
Main Generator Unscheduled Removal Transport Delay Time	0.60		
Main Generator Unscheduled Removal Maintenance Delay Time	0.70	0.10	0.10
Main Generator Unscheduled Removal Maintenance Preparation Time	4.	0.30	1.

ASSET EPGDS Method

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Inherent Availability

Maintenance Preparation Times(Flight Hours)

Main Generator Mean Time to Repair	3.190	
Main Generator Mean Maintenance Preparation Time	0.520	
Main Generator Mean Maintenance Down Time	3.710	
Main Generator Mean Time Between Maintenance	600.0	
Main Generator Inherent Availability	0.99385	

94

ASSET EPGDS Method

Figure 54

File		Run		Goto		Report		Help	
		Next		Previous		Back			
Airplane Parameters		NACELLE		EFGDS		Leading Edge		Floor Beam	
on		Weight		Configuration		Loads		Architecture	
Refueled takeoff speed		Generation		Distribution		System Attributes		Weight Summaries	
Stage Noise		db delta		AP acoustic level		Max GW / Eng		Max total thrust / Eng	
Thrust / GW ratio		Airplane Type		Number of Passengers		Range			

Airplane Parameters

Parameter	Value	Unit
ATA 24 Weight Summary	300000.	LB
FC 32 Weight Summary	2	KTS
Below Wing Weight	200.	
ATA 24 Weight Summary	3	

24-09, Electrical Power Distribution

24-10, Generator Drive

24-21, Power and Regulation

24-22, Controls and Indication

24-25, Back-up Generator

24-28, Feeders

24-31, Batteries

24-32, Transformer Rectifier

24-33, Emergency Generator

24-35, Flight-Control DC Power

24-40, External Power

24-51, AC Power Distribution

24-60, DC Power Distribution

VW-01, Wiring Provision

ATA Chapter 24 Weight Totals

150000.

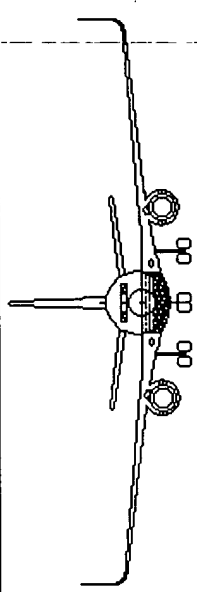
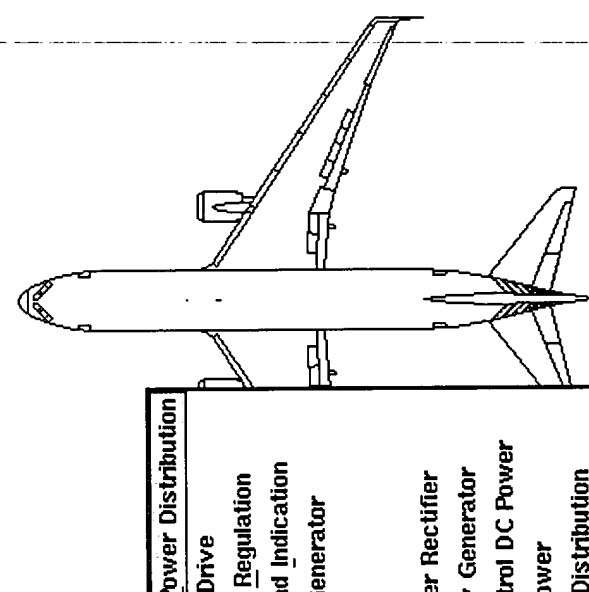
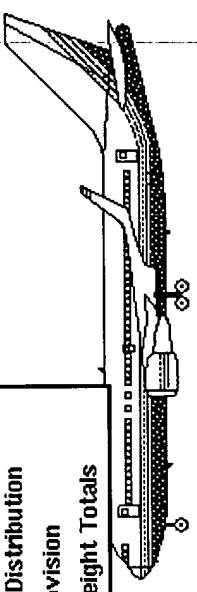
54450.

0.36

Narrow Body

221.

3370.

ASSET Main Module

FIGURE 55

24-09, Electrical Power Distribution

[illegible]

ASSET EPGDS Method

24-10, Generator Drive

[illegible]

ASSET EPGDS Method

Figure 57

24-21, Power and Regulation

[illegible]

ASSET EPGDS Method

24-22, Controls and Indication

[illegible]

ASSET EPGDS Method

Figure 59

Component

Component Designation

Quantity

Unit Wt

Subtotal

[illegible]

ATA 24-25, Back-up Generators

172.4 | B

ASSET EPGDS Method

24-28, Feeders

Component #		Component Designation		Quantity		Unit Wt		Subtotal	
<>	Gen_Fdrs	<>	Main Generator Feeders, R	<>	<>	68.8	LB <>	68.8	LB
<>	Gen_Fdrs	<>	Main Generator Feeders, L	<>	<>	68.8	LB <>	68.8	LB
<>	Gen_I	<>	Main Generator Feeders Installation, R	<>	<>	12.4	LB <>	12.4	LB
<>	Gen_I	<>	Main Generator Feeders Installation, L	<>	<>	12.4	LB <>	12.4	LB
<>	Gen_C	<>	Main Generator Feeders Connectors, R	<>	<>	10.3	LB <>	10.3	LB
<>	Gen_C	<>	Main Generator Feeders Connectors, L	<>	<>	10.3	LB <>	10.3	LB
<>	APU_Fdrs	<>	APU Feeders	<>	<>	68.8	LB <>	68.8	LB
<>	APU_C	<>	APU Feeders Connectors	<>	<>	12.4	LB <>	12.4	LB
<>	APU_I	<>	APU Feeders Installation	<>	<>	10.3	LB <>	10.3	LB
<>		<>		<>	<>	0.0	LB <>	0.0	LB
<>		<>		<>	<>	0.0	LB <>	0.0	LB
<>		<>		<>	<>	0.0	LB <>	0.0	LB
<>		<>		<>	<>	0.0	LB <>	0.0	LB
								274.4	LB
ATA 24-28, Feeders									

ASSET EPGDS Method

FIGURE 6)

24-31, Batteries

[illegible]

ASSET EPGDS Method

FIGURE 62

24-32, Transformer Rectifier

[illegible]

ASSET EPGDS Method

FIGURE 63

24-33, Emergency Generator

[illegible]

ASSET EPGDS Method

Figure 64

[illegible]

24-40, External Power

[illegible]

ASSET EPGDS Method

Figure 6b

24-51, AC Power Distribution

[illegible]

ASSET EPGDS Method

24-60, DC Power Distribution

[illegible]

ASSET EPGDS Method

FIGURE 68

WW-01, Wiring Provision

[illegible]

ASSET EPGDS Method

FILE 69

100

ATA 24-09, Electrical Power Distribution	655.1	LB
ATA 24-10, Generator Drive	113.6	LB
ATA 24-21, Power and Regulation	285.2	LB
ATA 24-22, Controls and Indication	15.0	LB
ATA 24-25, Back-up Generators	172.4	LB
ATA 24-28, Feeders	274.4	LB
ATA 24-31, Batteries	238.0	LB
ATA 24-32, Transformer Rectifier	64.4	LB
ATA 24-33, Emergency Generator	100.7	LB
ATA 24-35, Flight-Control DC Power	211.8	LB
ATA 24-40, External Power	59.5	LB
ATA 24-51, AC Power Distribution	106.0	LB
ATA 24-60, DC Power Distribution	49.4	LB
WW-01, Wiring Provision	152.6	LB
Electrical Power Generation & Distribution System	2498.0	LB

ASSET EPGDS Method

Figure 70

Airplane Parameters

Airplane application

Maximum Takeoff Weight

3000000.

LB

FC 32 Report

Component #	Component Designation	Qty	Unit	Wt Su (LB)
32	Electrical Power Generation & Distribution System			
32-01	AC Power System	1		255.3
32-01-01	AC POWER GENERATION EQUIPMENT	1		233.9
32-01-01-01	MAIN AC POWER GENERATORS INSTLD	2		90.0
32-01-01-01-01	PRIME DRIVE GENERATOR	2		4.8
32-01-01-01-02	QUICK ATTACH DETACH (QAD)	2		13.8
32-01-01-01-03	GENERATOR FLUIDS	2		4.7
32-01-01-01-05	HARDWARE INSTALLATION	2		3.6
32-01-01-01-06	WIRING INSTALLATION	1		5.0
32-01-01-02	GENERATOR CONTROL UNITS	1		16.4
32-01-01-06	BUS POWER CONTROL UNITS	1		269.6
32-01-05	EROPS-VSCF POWER GENERATION SYSTEM	1		95.9
32-01-05-01	VSCF GENERATORS & OIL	2		38.1
32-01-05-01-01	VSCF GENERATOR	2		9.9
32-01-05-01-02	VSCF GENERATOR OIL			

Return

send to printer

save to file

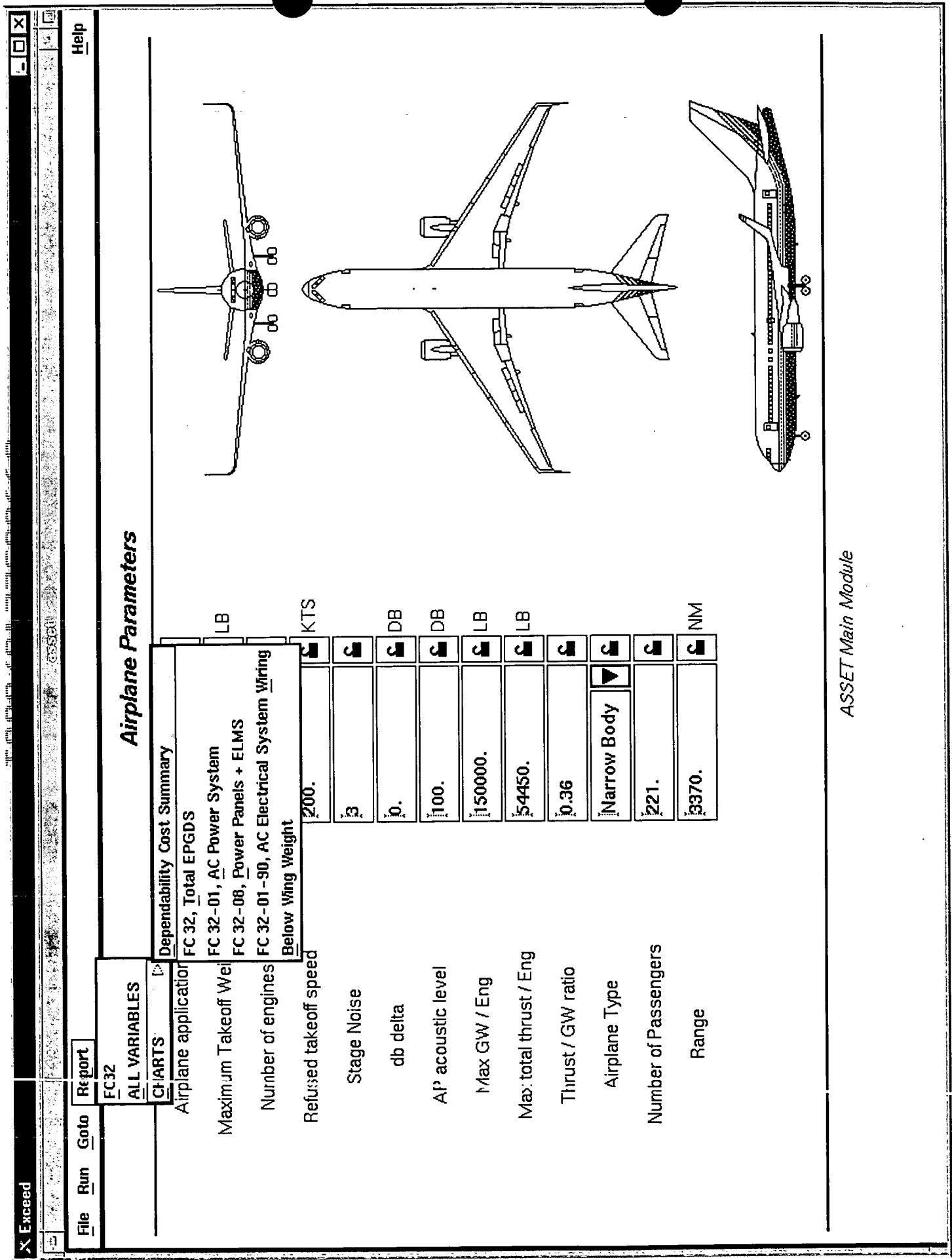
ASSET Main Module

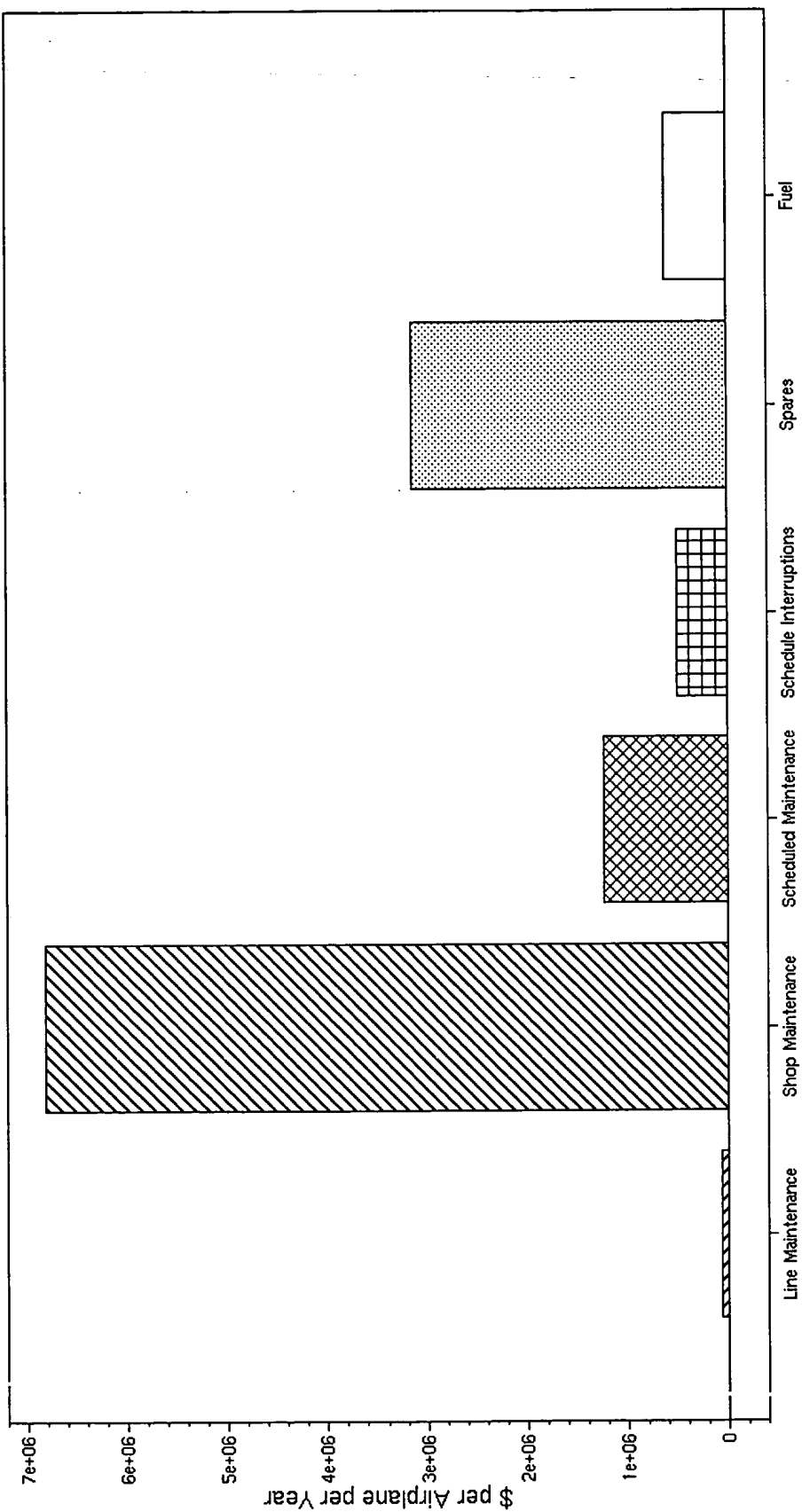
Figure 73

Airplane Parameters

Airplane application		300000.		LB					
Maxim um Takeoff Weight									
Airplane application									
Maxim um Takeoff Weight									
Airplane application									
Maxim um Takeoff Weight									
Airplane application									
Maxim um Takeoff Weight									
Airplane application									
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FIGURE 74



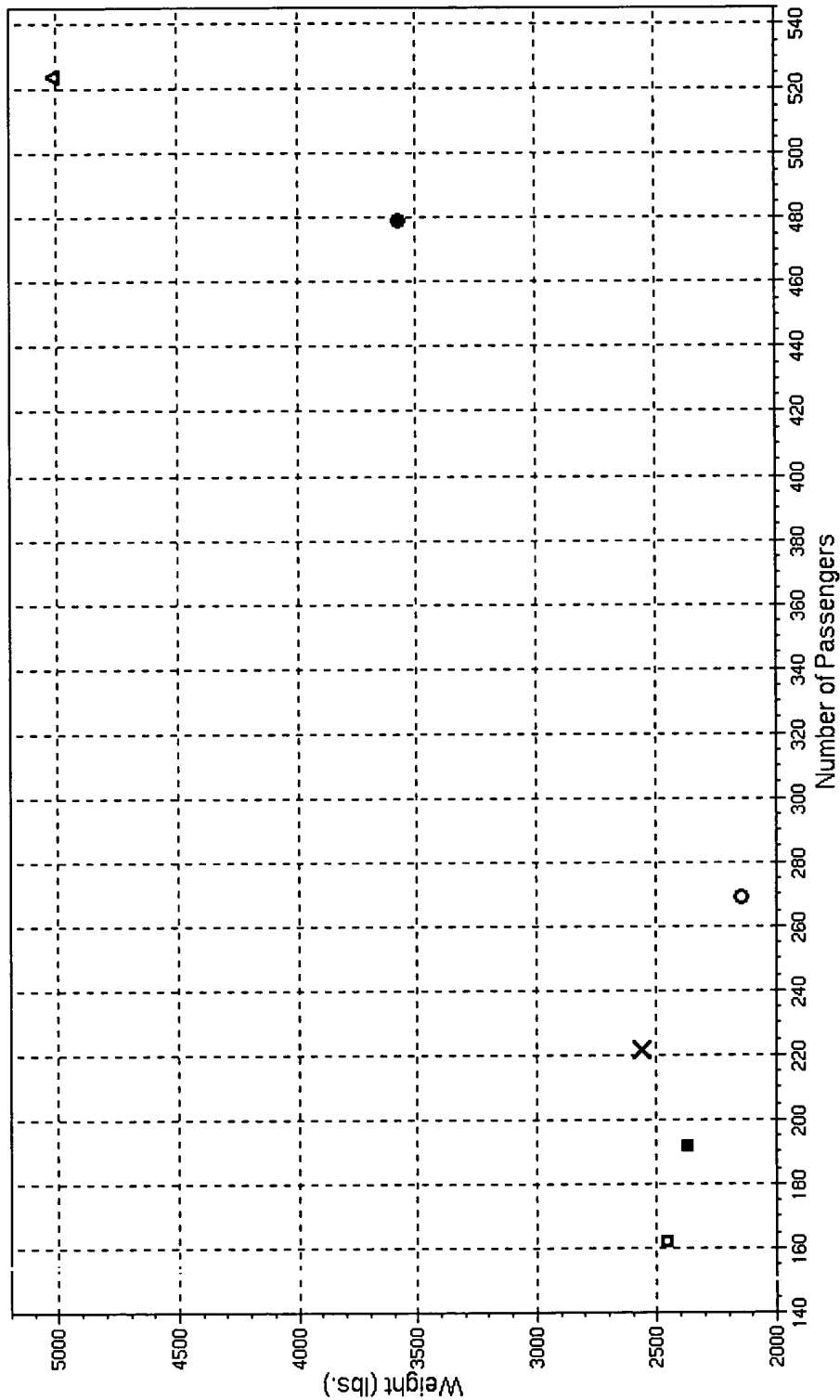


100

Return

Figure 7b

FC32, Total EPGDS Weight



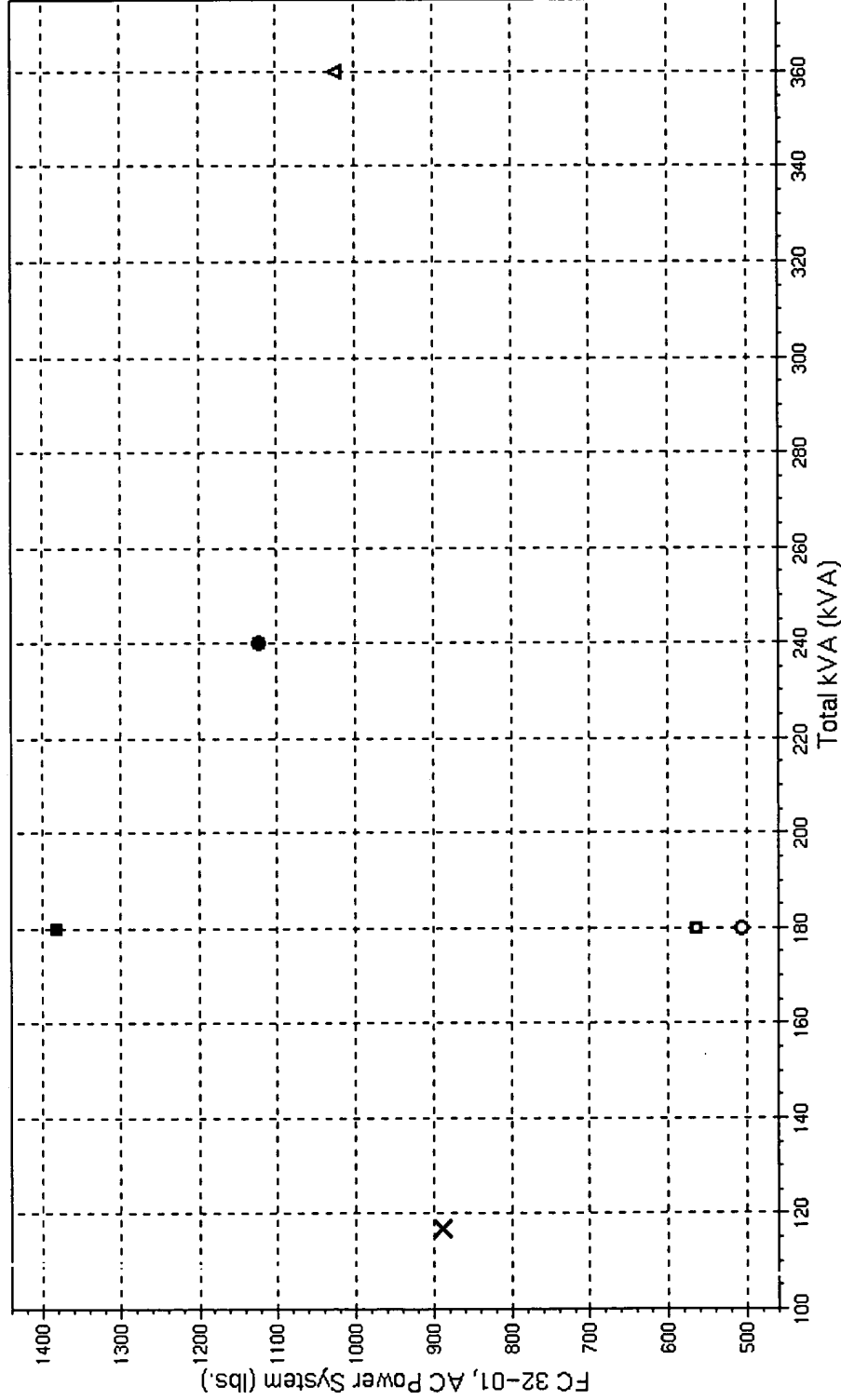
102

Return

Figure 77

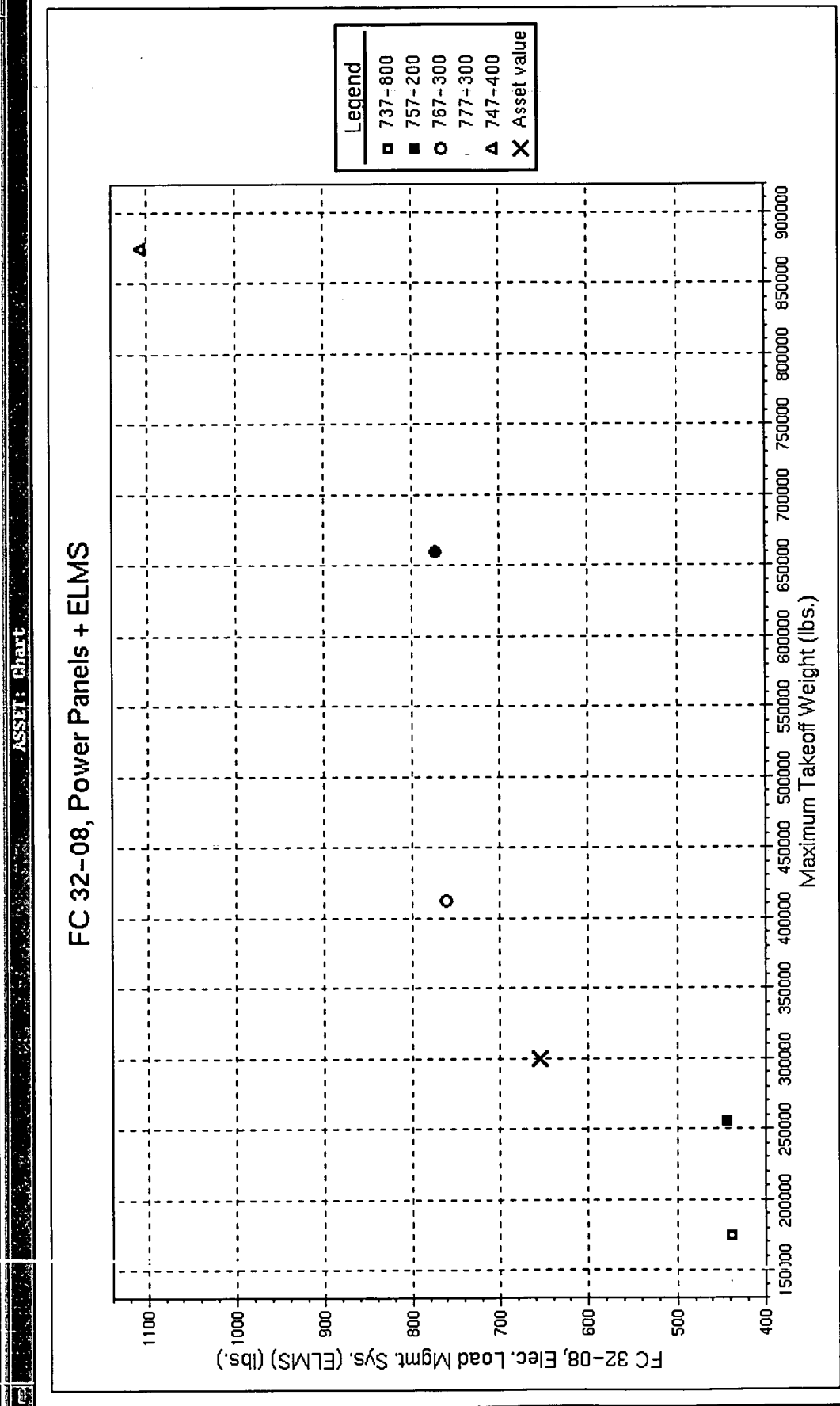
ASST: Chart

FC 32-01, AC Power System Weight



Return

104

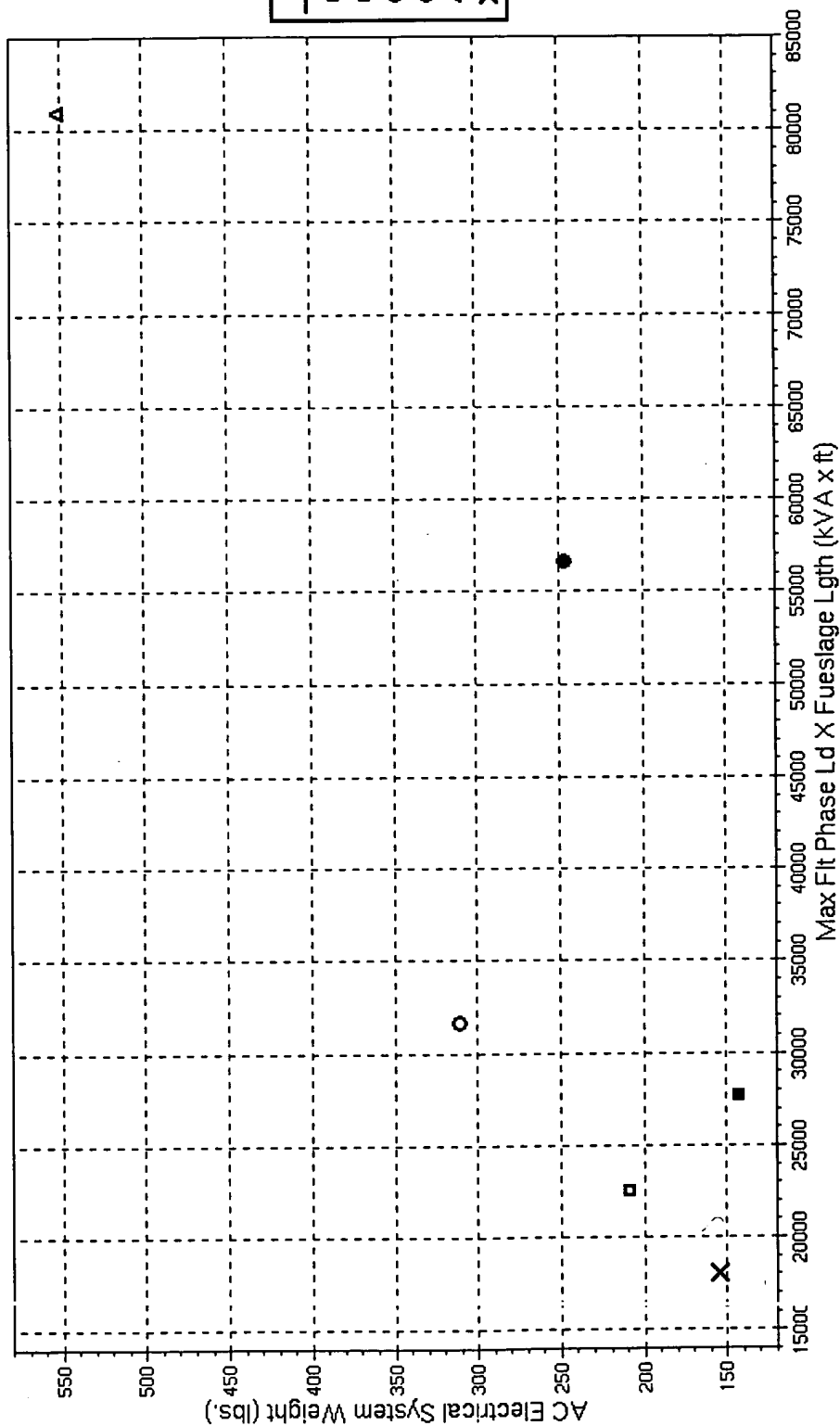


Return

106

Figure 79

FC 32-01-90, AC Electrical System Wiring

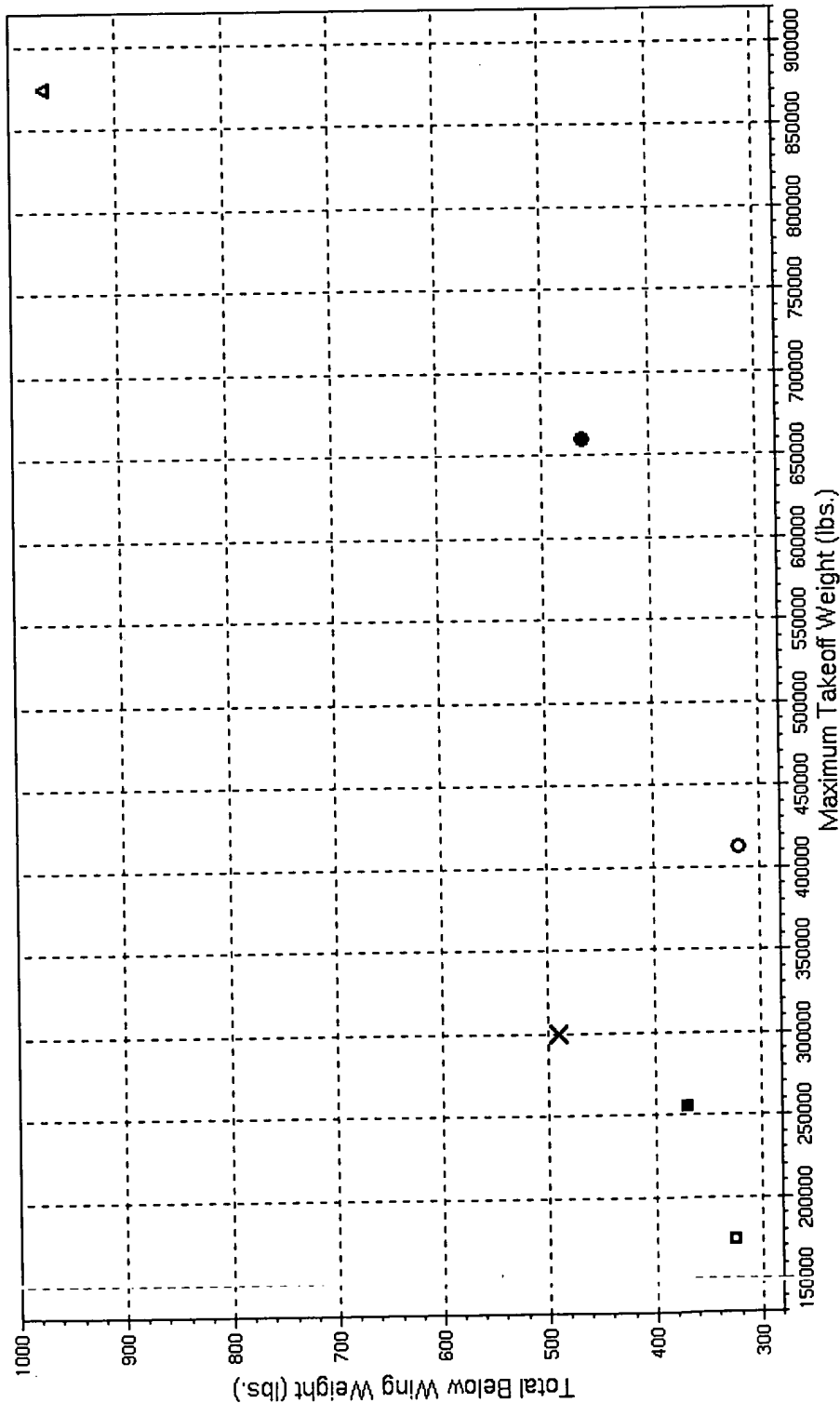


Return

108

ASSET: Chart

Below Wing Weight



Return

110

Figure 18